

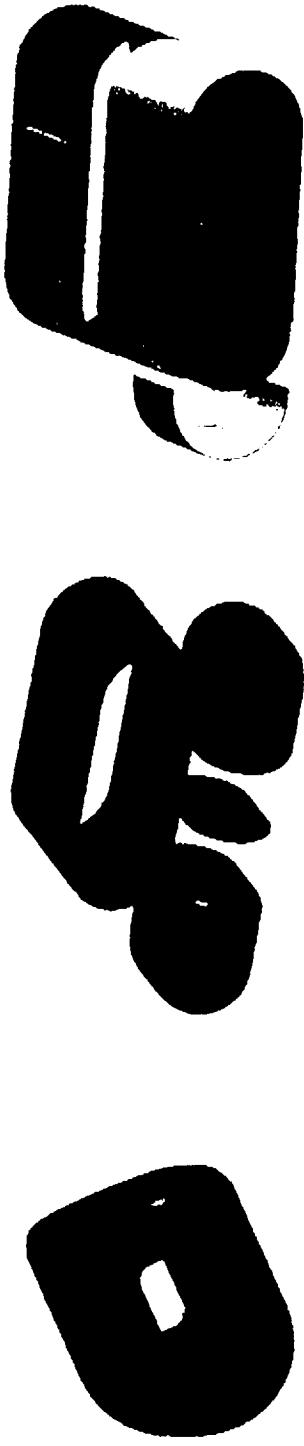
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AN INTEGRATED APPROACH TO PROPULSION



Composite, Cryogenic, Conformal, Common Bulkhead, Aerogel-insulated Tank (CBAT)

Materials and Processing Methodologies



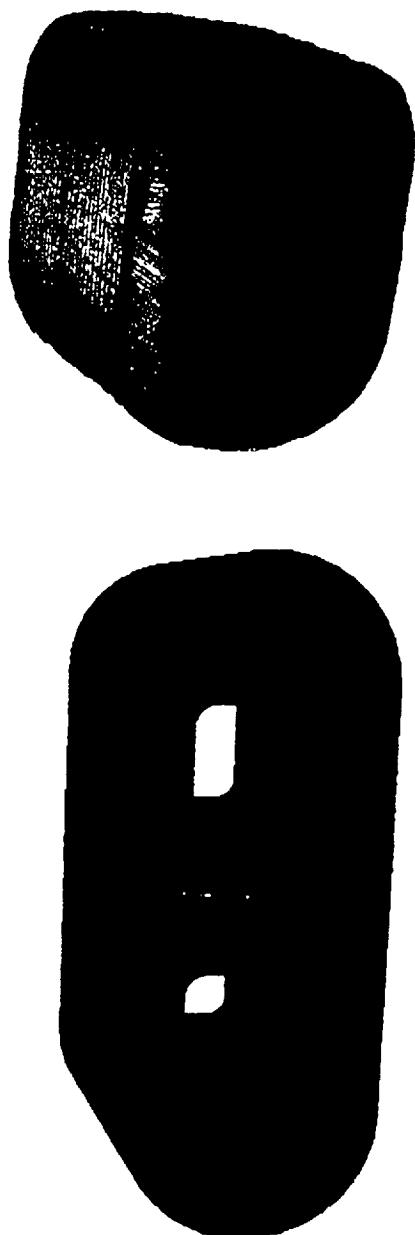
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Michael P. Kovach
September 19, 2000

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Agenda

- What is CBAT?
- Current Status
- Materials & Processes
 - Fabrication Method
 - Challenges
- Future Activities
- Summary





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Aerojet Thiokol Propulsion
Aerojet Rocketdyne, Inc.

Introduction

What is CBAT?

- Program to evaluate new technologies for future generation Reusable Launch Vehicles (RLVs).

• Accomplished through:

- Design
- Fabrication
- Testing of subscale system

• CBAT stands for:

- Composite
- Cryogenic
- Conformal
- Common Bulkhead
- Aerogel-insulated
- Tank



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Alaris

Introduction

CBAT - Composite

- **Composite refers to:**
 - Polymer Matrix Composites (PMCs) used in fabrication.
- **PMCs**
 - Materials of choice for most of today's aerospace applications.
 - High Strength/Stiffness
 - Low Density
 - As well as:
 - Resistant to fracture, corrosion, and wear.
 - Allow near net shape fabrication
 - Facilitate component integration
- **CBAT will use graphite/epoxy prepeg tow and fabric.**
 - Hexcel IM7 (12k and 5HS)
 - Cytec-Fiberite 977-2

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Introduction

CBAT - Cryogenic

- Cryogenic refers to:
 - Prospective propellants for use with the system.
- Cryogenic propellants
 - High energy to mass
 - Challenges:
 - Reactivity
 - Thermally-induced strains
- CBAT design is based on LOX/RP system.
 - Testing will use LN₂ instead of LOX.
 - Increased Safety
 - Reduced Cost
 - RP may be replaced by LH₂.

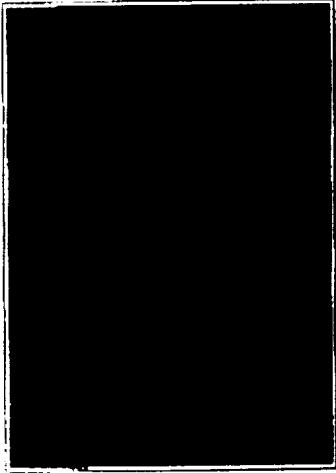
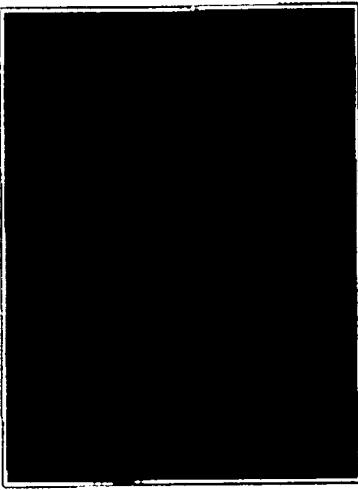
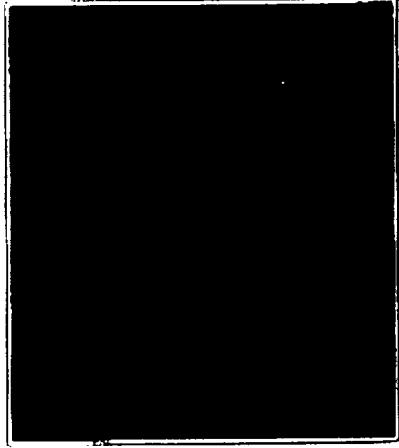
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Introduction

CBAT - Conformal

- **Conformal refers to:**
 - Capability to customize the propellant storage system to the shape of the vehicle it will be used in.
- **Conformal Storage Systems**
 - Maximize capacity
 - Facilitate integration into vehicle structure
- **CBAT design simulates a conformal-type system which could be used in conjunction with future generation RLVs.**



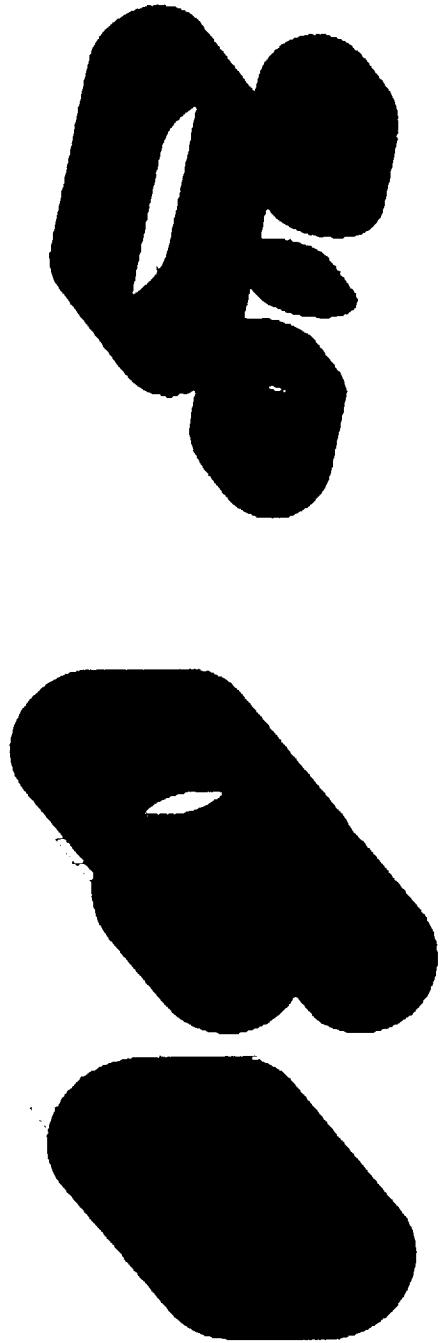
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Thiokol Propulsion
A Division of Thiokol Corporation
A Thiokol Company
Propulsion Systems
Propulsion Components
Propulsion Components

Introduction

CBAT - Common Bulkhead

- **Common Bulkhead refers to:**
 - Geometry or methodology for placing/packaging the propellant storage tanks adjacent to one another within the vehicle
- **Common Bulkhead Propellant Storage Systems**
 - Reduce design restrictions
 - Maximize capacity
- **CBAT design simulates a conformal-type system which could be used in conjunction with future generation RLVs.**



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Thiokol is a registered trademark of Thiokol Corporation.

Introduction

CBAT - Aerogel-insulated

- **Aerogel-insulated refers to:**
 - Use of a special class of materials to insulate around and between certain portions of the system.
- **Aerogels**
 - Low density
 - Low thermal conductivity
- **CBAT will potentially use aerogels in various forms.**
 - Particulate
 - Thermal blankets
 - Encapsulated packets
 - Monolithic



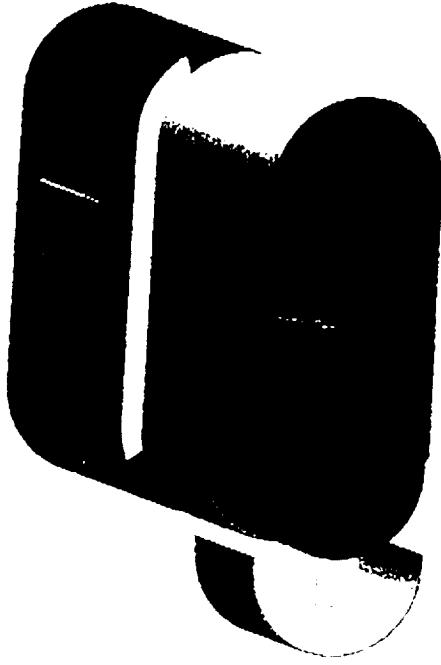
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ALTA

Current Status

CBAT - Design, Fabrication, & Test

- **Design**
 - Incorporates each of the aforementioned technologies and/or functionalities.
 - Component form and fit have been verified using rapid prototyping.
 - Stereolithography Apparatus (SLA)
 - Laminated Object Manufacture (LOM)
- **Fabrication**
 - Fabrication of tooling, fixtures, and metal hardware is continuing.
 - Winding mandrel molding is complete (1st mandrel)
 - Material characterization and design verification specimens are being processed.
- **Test**
 - Material characterization testing started in April 2000.
 - Projected start of component testing is October 2000.



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AN INTEGRATED APPROACH TO PROPULSION SYSTEMS

Materials & Processing

Fabrication Method

- Tanks will be fabricated using a combination of processes.
 - Filament Winding
 - Removable mandrel
 - Hand Lay-up
- Remaining components will be produced by hand lay-up.
- Adhesive will be used to bond the tanks, common bulkhead assembly, and skirt inner skin.
- Outer skin and core will be hand laid-up on the assembly and co-cured/bonded into place.



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Propulsion Systems for Space, Defense, and Commercial Applications

Materials & Processing

Challenges

- Potential challenges addressed to date:
 - Filament Winding
 - Hand Lay-up
 - Tooling
- Remedied through use of:
 - Subscale Models
 - Simulation
 - Test

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Propulsion Systems Division
Propulsion Systems Division
Propulsion Systems Division

Materials & Processing

Challenges - Filament Winding

- **Problem(s)**
 - Build-up in the dome regions.
 - Tow slippage.
- **Approach**
 - Evaluate wind options
 - Polar/Hoop
 - Helical
 - Wind subscale models (1/5 scale).
 - Use Rapid Prototyping for the mandrels.
 - Enhance Tack
 - Increase solvent content of tow.
 - Place film adhesive on mandrel prior to winding.

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Materials & Processing

Challenges - Hand Lay-up

- Problem(s)
 - Lay-up of prepreg cloth in dome regions.
 - Fiber distortion
 - Wrinkling
- Approach
 - Drape testing
 - Quantify distortion
 - Simulation
 - Create ply patterns with minimal chance for wrinkling
 - Factor fiber distortion into structural models.

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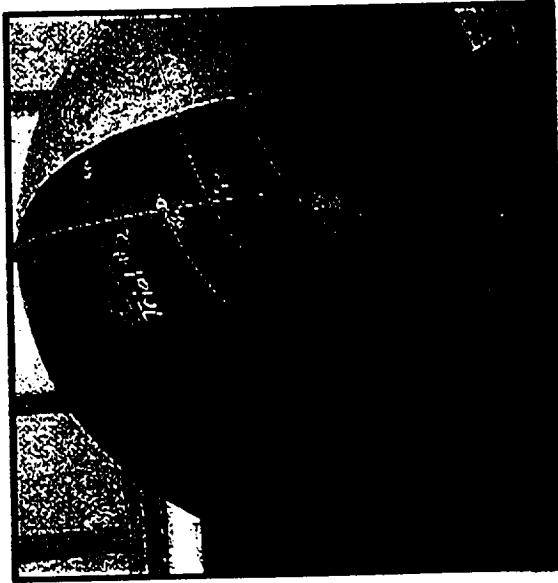
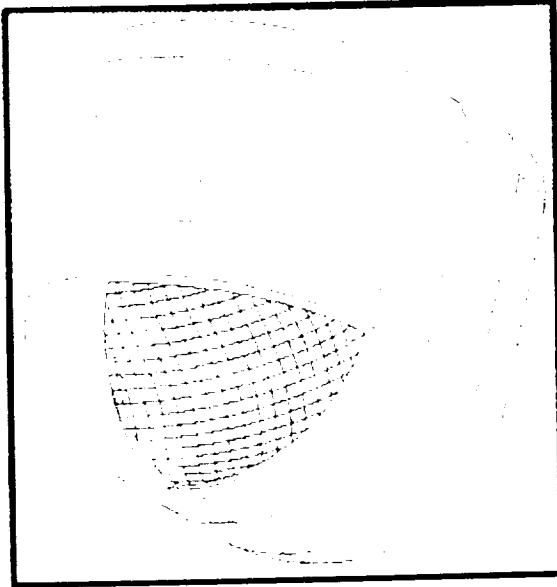
Advanced Propulsion System Development

Materials & Processing

Challenges - Hand Lay-up

• Results

- Drape testing quantified distortion for use in simulation.
- Simulation modeled plies
 - Generated ply patterns to minimize wrinkling and distortion
 - Provided fiber distortion data to feed into the structural model.

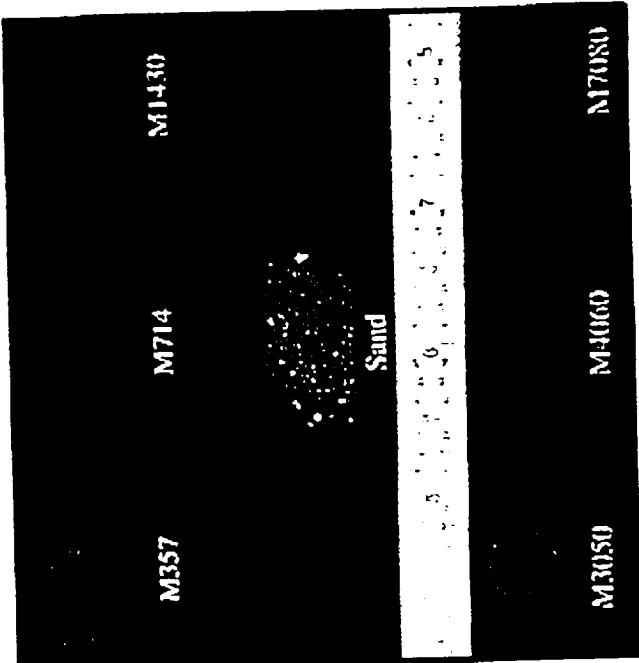


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Materials & Processing

Challenges - Tooling

- Problem(s)
 - Previously used Macrolite™ binder incompatible with 350°F cure.
- Approach
 - Use sand mandrel binder, sodium silicate
 - Proven temperature capability.
 - Evaluate on subscale articles.
 - 5.75-inch mandrel
 - 8-inch wound vessel



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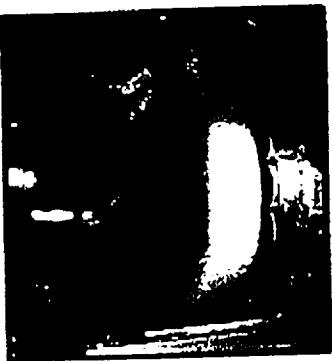
Propulsion Components Division
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Materials & Processing

Challenges - Tooling

Results

- Sodium silicate binder with original Macrolite™ formulation formed insoluble mandrel.
 - Fine particles enhanced packing, eliminated porosity, and prevented water intrusion during washout.
- Replacement of finest Macrolite™ particles and alumina microspheres with sand, and minimization of the binder resulted in a mandrel with good washout characteristics.
 - 3% heavier than a mandrel made using original Macrolite™ formulation
 - 65% lighter than a mandrel made using sand



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Future Activities

- **Near Term**
 - Fabrication
 - First Full Scale Winding (Tank Shell) for door test
 - Scheduled Completion - 9/29/00
 - Tie Down Skirt for door test
 - Scheduled Completion - 10/6/00
 - Two doors for door test
 - Scheduled Completion - 10/6/00
 - Second Full Scale Winding for sectioning and coupon testing.
 - Scheduled Completion - 11/10/00
 - Test
 - Full Scale Door Seal Test
 - Scheduled for 10/00
- **Long Term**
 - Fabricate and conduct cyclic testing of Full Tank System Assembly
 - Scheduled completion 9/01
 - Install tank system TPS and conduct LH₂/LN₂ structural test.
 - Scheduled completion 9/02

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Summary

- CBAT is in the process of Designing, Fabricating, and Testing a propellant storage system that will integrate various technologies and functionalities.
 - Composites
 - Cryogenic propellants
 - Conformal shape
 - Common Bulkhead
 - Aerogel insulation
- Program is still in the fabrication phase.
- Following completion of fabrication and test, concept could be incorporated into future generation RLVs.
 - Improve performance and reliability
 - Reduce weight